



Aircraft Catastrophic Failure Prevention Program

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Uncontained Engine Failure Research

Sioux City, 1989



Figure 3.--Photo (C. Zellmer) taken while flight 232 was approaching Sioux Gateway Airport. Arrows indicate damage to the right horizontal stabilizer. It is also evident that the No. 2 engine fan cowl door and the tail cone are missing.



Uncontained Engine Failure Research Fan Disk, 1989

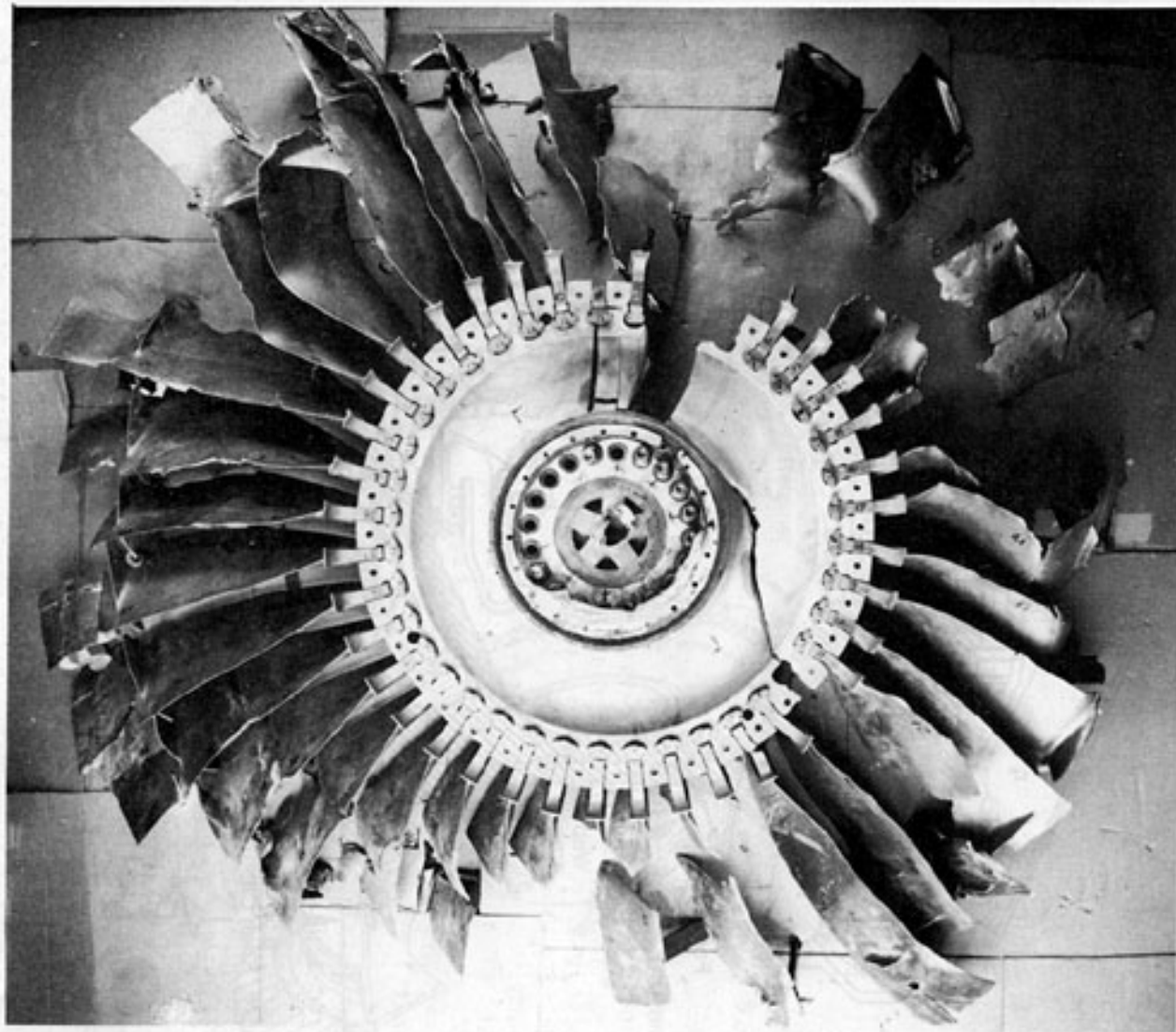
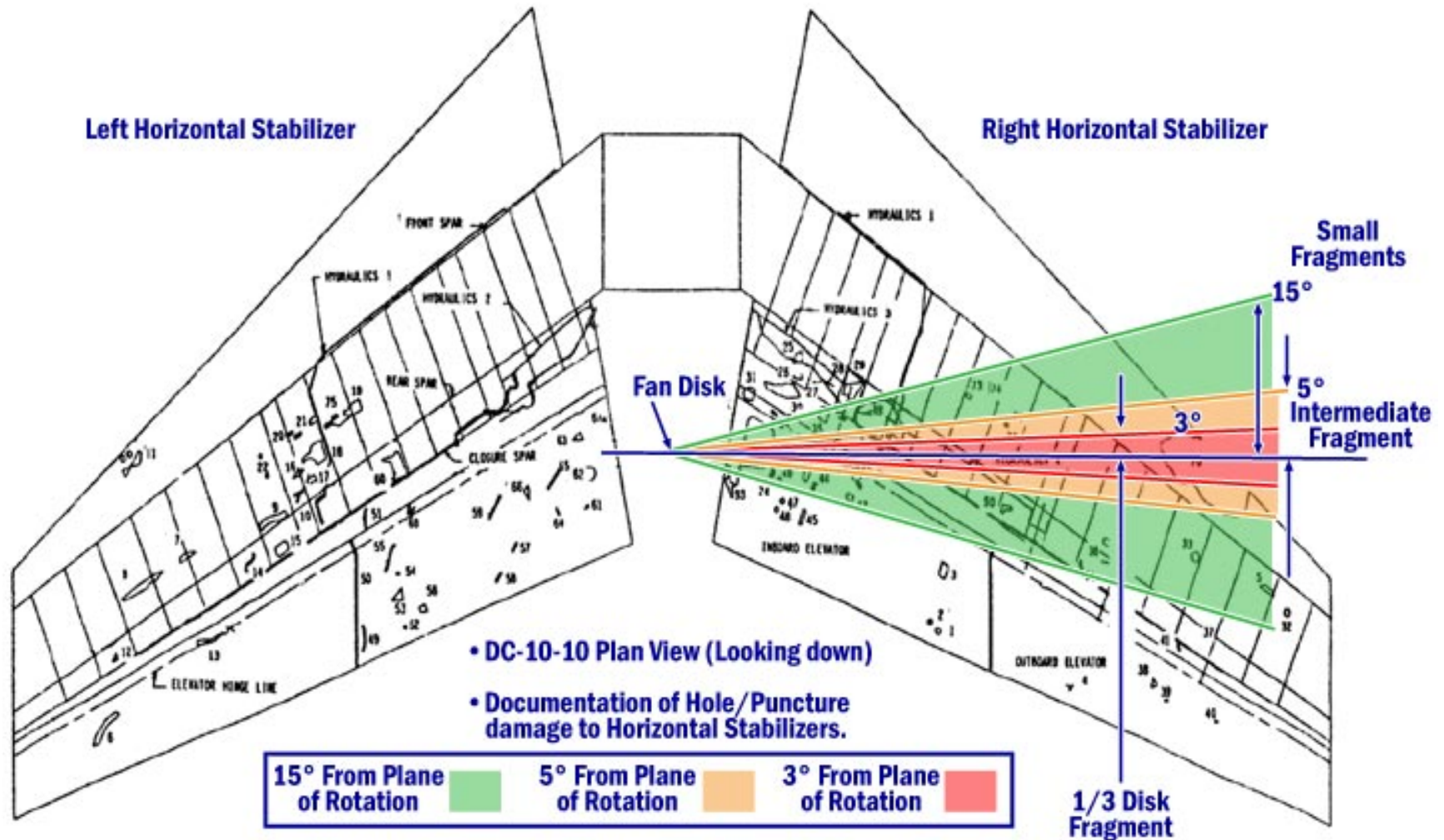


Figure 18.--No. 2 engine stage 1 fan disk (reconstructed with blades).



Uncontained Engine Debris





NTSB Findings- SIOUX CITY 1989

- ❖ #2 (center) engine uncontained failure
 - Compressor disk material defect in manufacturing
 - Maintenance Inspection of compressor disc

- ❖ Uncontained debris from engine damaged all three aircraft hydraulic systems



FAA R&D Program Background

Aircraft Catastrophic Failure Prevention Program (ACFPP)

- ❖ **ACFPP was created by Congressional direction after the 1989 Sioux City Accident**
- ❖ **Objective-**
 - **Conduct research that will reduce the risk of catastrophic aircraft accidents and fatalities**
- ❖ **Uncontained engine failure research has been primary focus**
 - **Mitigation of smaller fragments**
- ❖ **Other research programs concentrated on disc material and inspection improvements**



Uncontained Engine Debris Mitigation Program

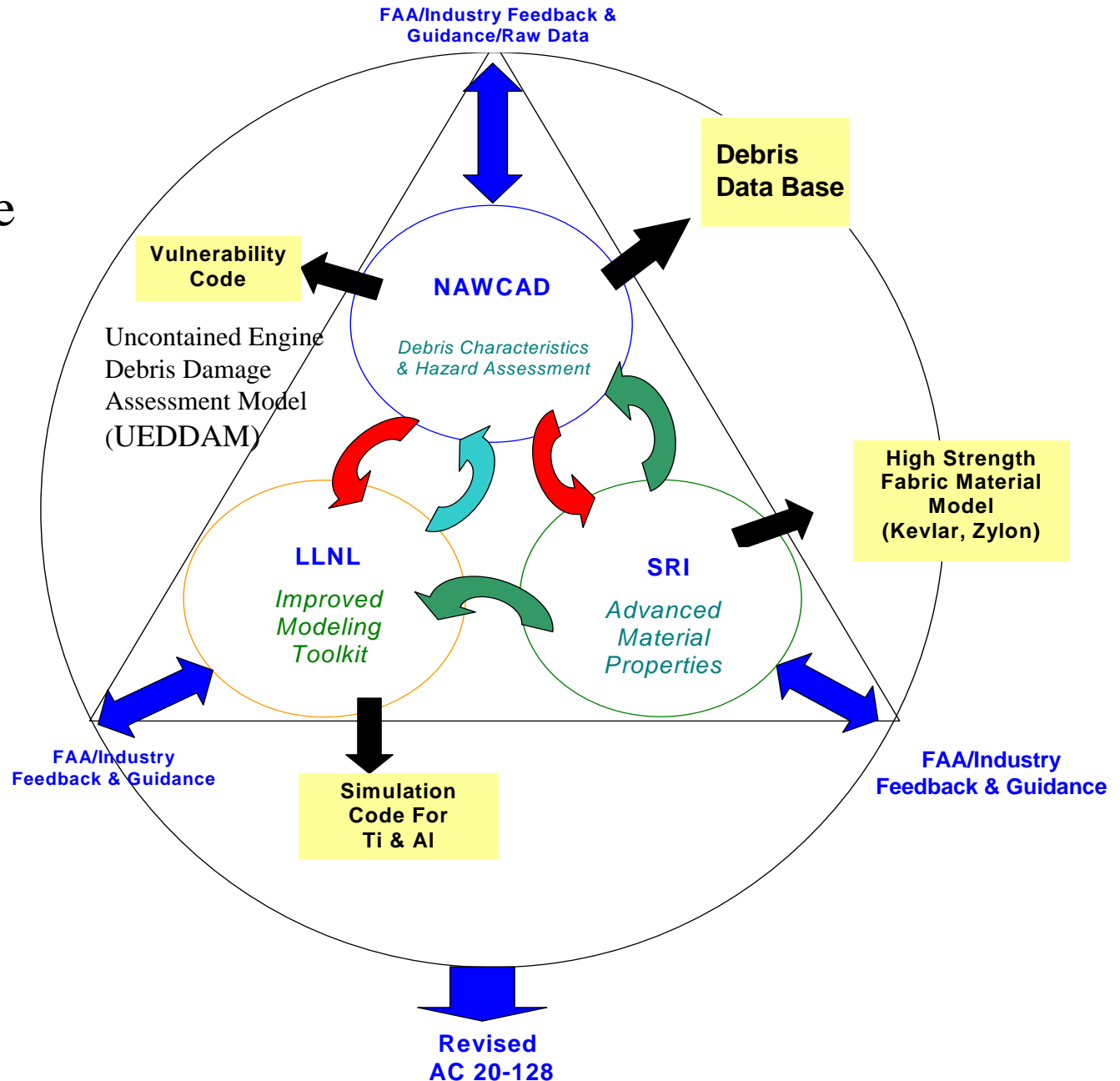
- ❖ NTSB Recommendations from Sioux City
 - *Aircraft Vulnerability*- Develop and maintain a database of uncontained engine debris impacting aircraft that would benefit design assessments and safety analysis
 - *Failure Mitigation*- Update AC-20-128 to mitigate uncontained engine debris from new aircraft certification
 - multiple fragments
 - Non-linear finite element modeling



RPD 516 - Uncontainment Research: Airplane Focus

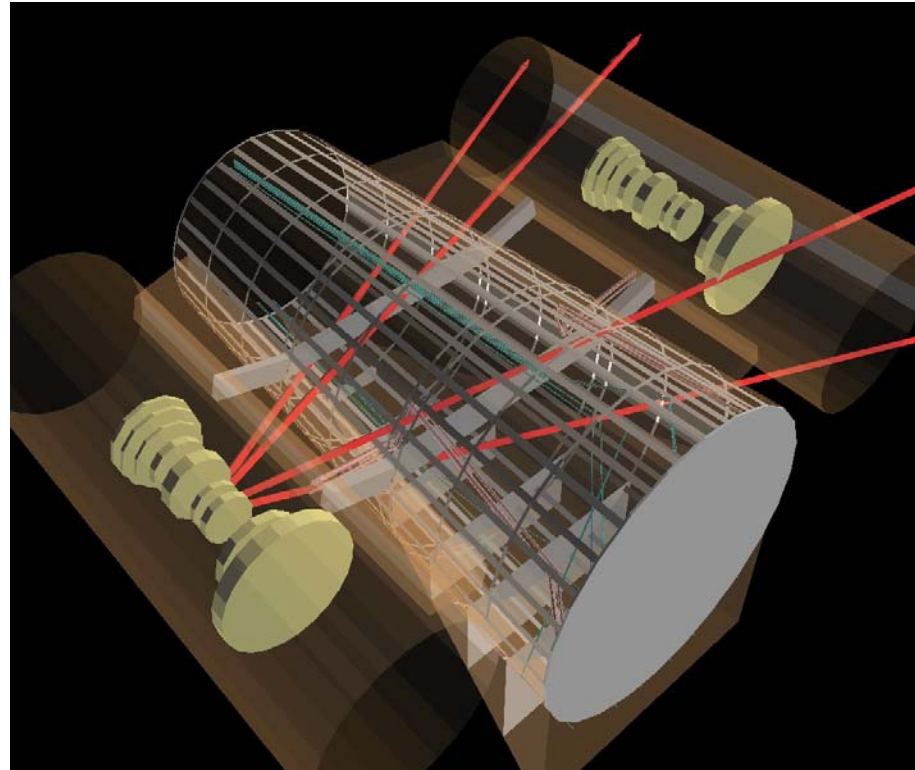
Uncontained Engine Debris Mitigation Program

- NTSB A 90-170, revise AC-20-128 (new aircraft designs)
- NTSB A 90-172, Develop data base of uncontained debris





Uncontained Engine Debris Damage Assessment Model (UEDDAM) Trajectories



UEDDAM model repeats event multiple times.
Varies trajectories and orientation each time.



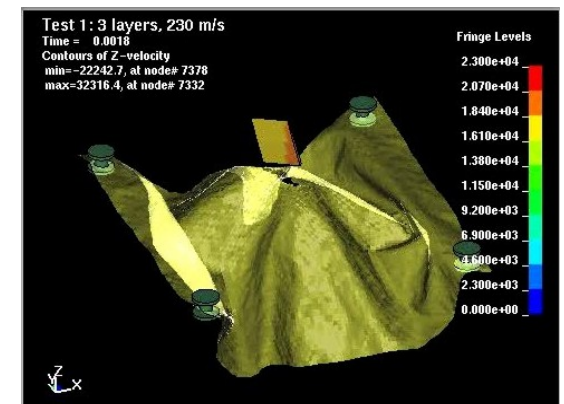
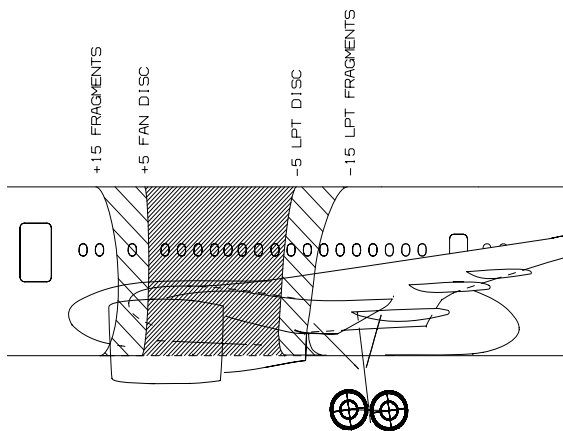
AACE University / Industry Partnerships

- ❖ 2001 Uncontained Engine Workshop presented results of FAA sponsored research
- ❖ In Order to Transition this Technology into service, partnerships were formed between academia, government, and industry using AACE Cooperative Agreements
- ❖ 100 Percent Cost matching achieved (*primarily by industry partners*)



AACE University / Industry Partnerships (2002 start)

- UC Berkeley partnered with Boeing, SRI International and Lawrence Livermore National Laboratory (LLNL)
- *Research Area* - material development and modeling for aircraft barriers (metals, fabrics, composites)



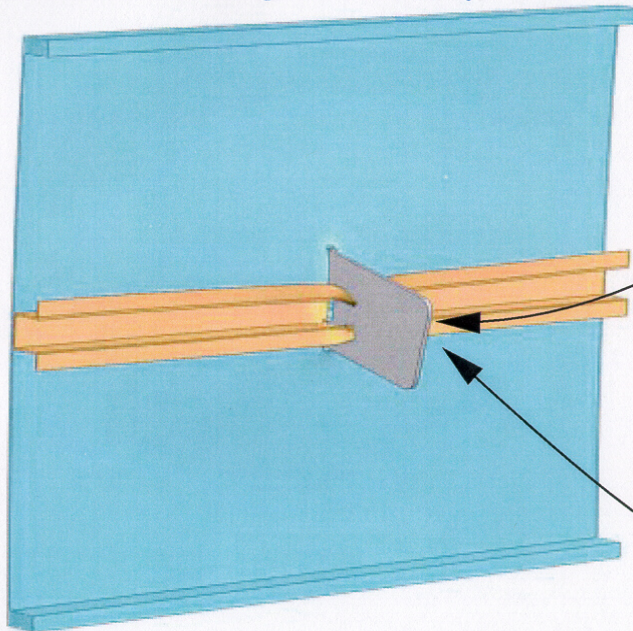


AACE Technology Transfer Engine Fragment Shielding Project

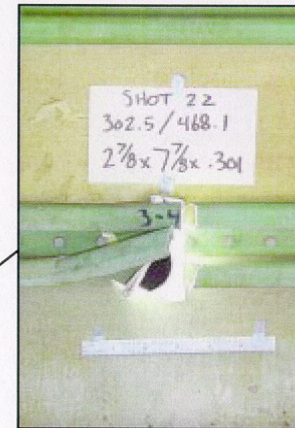
New 2024-T3 Aluminum failure parameters used to simulate Chinalake Engine Debris Fuselage Penetration Testing Phase 1: Test 22 – Fan blade fragment impact with skin/hat section



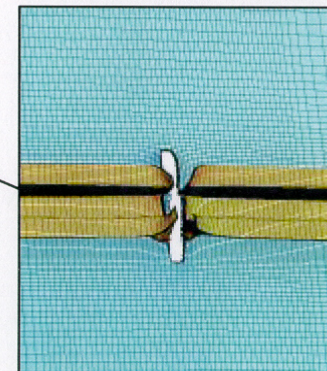
Initial blade fragment velocity = 795 ft/sec



The mesh contained 122,625 nodes and 94,420 elements. The rivets were not modeled in the simulation.



Residual fragment velocity = 725 ft/sec

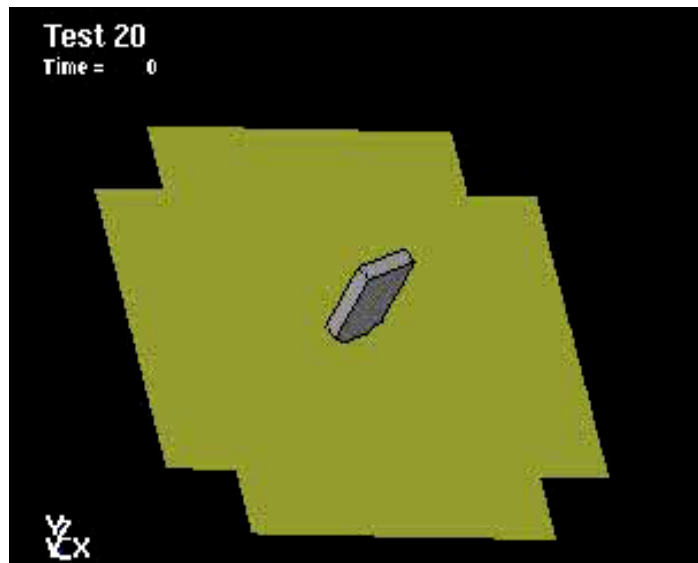


Residual velocity = 715.833 ft/sec
-1.26% deviation from the measured value.

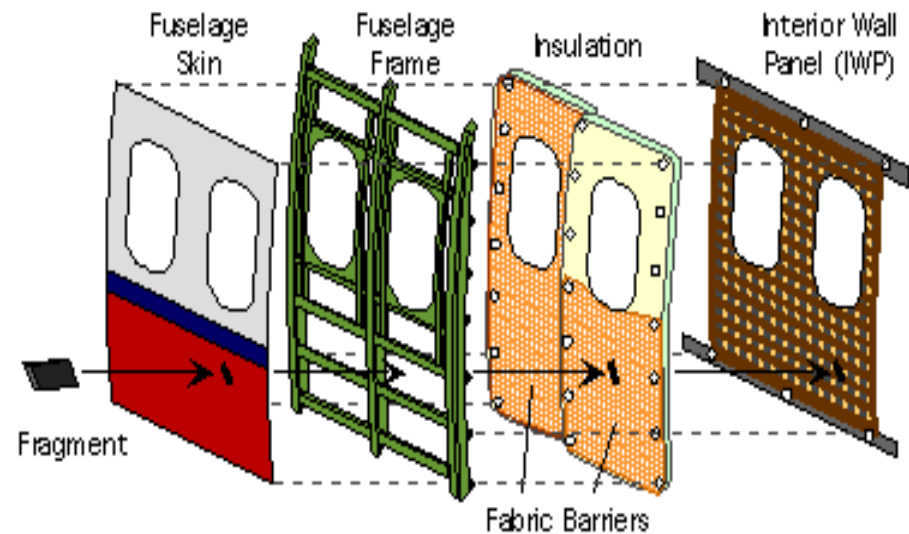


AACE University / Industry Partnerships (cont'd)

Simulation



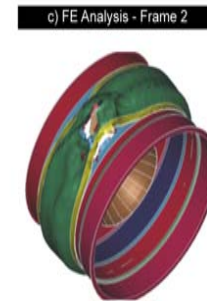
Barrier Concept





AACE University / Industry Partnerships (2002 start)

- **Arizona State University** partnered with Honeywell , SRI International and NASA Glenn
- *Research Area-* turbine engine fabric containment modeling



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LESSONS LEARNED

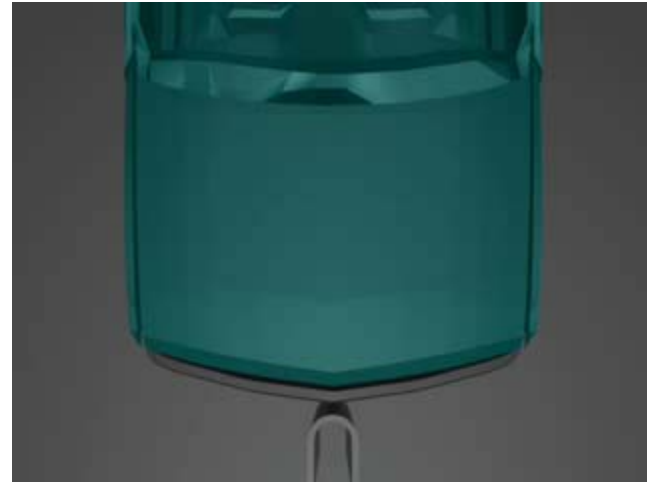
FROM ASU / UCB PROGRAMS

- ❖ **LS-DYNA can give significantly different results based on Model version and computer platform**
- ❖ **LS-DYNA Model has extensive Quality Control Program for the Automotive Industry**
 - **No such Quality Control Program existed for the Aircraft Industry**
- ❖ **FY 2003, a Joint FAA / NASA sponsored Workshop established an Aerospace Quality Control Working Group with Industry (Engine / Airframe Manufacturers)**



AACE University / Industry Partnerships (2004 start)

- George Washington University partnered with Livermore Software Technology Corp (LSTC) and Silicon Graphics Inc. (SGI)
- *Research Area-* LS-DYNA Aerospace Working Group Quality Control Support
- National Crash Analysis Center: www.ncac.gwu.edu





AACE Technology Transfer Engine Fragment Shielding Project

- ❖ ***“Modeling, Analysis and Testing of Metallic and Composite Shielding”***
- ❖ AACE Project started in FY-02 (Phase 1)
- ❖ **Team: UC Berkeley (Prof T. Zohdi PI), Boeing, LLNL**
 - ***Purpose-*** To develop an accurate LS DYNA3D model for aircraft materials and barriers using metals, composites, etc.
 - ***Accomplishments (Phase 1)(FY-02-04)***
 - Completed ballistic testing on aluminum at UCB and LLNL
 - Boeing modeled ballistic tests using Livermore improved material model
 - ***Status for FY-04 (phase 2)(FY-04-06)***
 - Composite testing and modeling is prime focus. Titanium will also be evaluated for thick plates. Ballistic and material testing is currently in progress at UC Berkeley and LLNL.

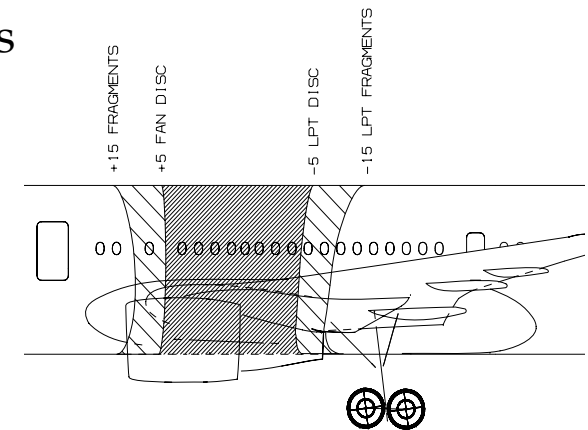
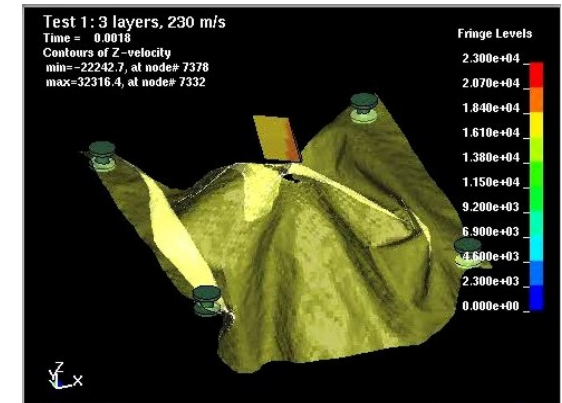




Technology Transfer - Lightweight Ballistic Protection on Commercial Aircraft

❖ **Team:** UCB, (Prof. T. Zohdi PI), Boeing, SRI International

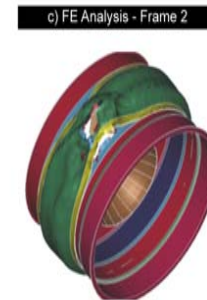
- **Purpose-** Determine the suitability of Zylon fabric as a barrier against uncontained aircraft engine fragments.
- **Accomplishments (Phase 1)- (FY-02 /03)**
 - Completed ballistic testing at UCB and SRI to determine material characteristics
 - LS DYNA material model developed by SRI used by Boeing to predict ballistic test results
 - Completed independent Zylon aircraft compatibility material testing at Boeing
 - DOT/FAA/AR-04/40,P1-3 report
- **Status Aircraft Testing (Phase 2)-(FY-04-05)**
 - Optimize aircraft attachments designs on small scale ballistic testing at UCB
 - Conduct full-scale fuselage testing at NAWC China Lake with Boeing designed fabric shields. Report in progress.





AACE Technology Transfer Multi-Layer Fabric Engine Containment Project

- ❖ Team: ASU (Prof. S Rajan PI), Honeywell, SRI, NASA-Glenn
 - *Purpose-* To develop an accurate LS DYNA3D model for engine containment systems
 - *Accomplishments (Phase 1) -*
 - Completed static testing at ASU and SRI
 - Completed ballistic testing at NASA Glenn as part of NASA Engine Containment Program
 - Completed computational analysis at Honeywell and SRI of ballistic testing with excellent correlation.
 - DOT/FAA/AR-04/40,P1-4 report
 - *Status (phase 2)*
 - This phase will refine ballistic testing to more represent engine blade-out condition
 - Containment ring testing is currently underway at NASA Glenn





FAA Certification Support LS-DYNA Modeling Quality Control

- ❖ New AACE Project started in FY-05
- ❖ Team: George Washington University (GWU) (Dr. S. Kan PI), Livermore Software Technology Corp (LSTC), Silicon Graphics Inc. (SGI)
- ❖ Purpose- Provides more accurate modeling of impact and penetration for engine containment and fuselage barrier designs.
 - Provide direct support to the LS-DYNA Aerospace Quality Control Working Group
 - Direct benefit to FAA Certification by establishing modeling standards
 - Provide LS-DYNA training for FAA and industry

